



## University of Nova Gorica Graduate school, Physics

# Turbulence modeling and its applications Seminar

Matej Andrejašič

Ajdovščina, 11th January 2010

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## Laminar flow

- streamlines are smooth and regular,

## Turbulent flow

- streamlines break up,
- fluid elements move in a random, irregular and torous fashion.

## Turbulence modeling:

- mathematical model that approximates the physical behavior of turbulent flow.
- much simpler than the full time dependent Navier-Stokes equations,
- complex enough to capture the essence of the relevant physics.



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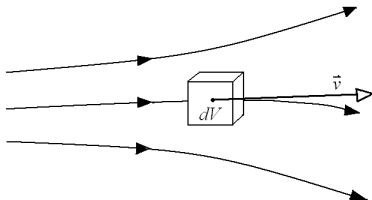
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# Flow model

Infinitesimally small moving fluid element of fixed mass.



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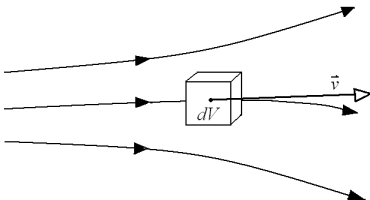
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Infinitesimally small moving fluid element of fixed mass.



Newton's second law

$$\frac{d(m \mathbf{v})}{dt} = \mathbf{F}$$

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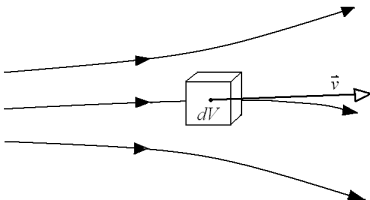
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Newton's second law

$$\frac{d(m \mathbf{v})}{dt} = \mathbf{F}$$

Navier-Stokes equation

$$\rho \frac{\partial u_j}{\partial t} + \rho U_j \frac{\partial u_j}{\partial x_j} =$$

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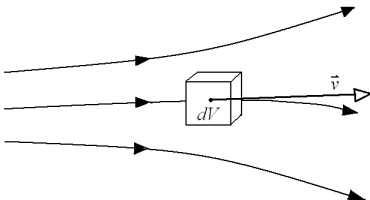
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$$\rho \frac{\partial u_i}{\partial t} + \rho U_j \frac{\partial u_i}{\partial x_j} = -\frac{\partial p}{\partial x_i} +$$

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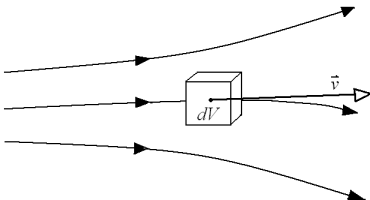
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$$\rho \frac{\partial u_i}{\partial t} + \rho U_j \frac{\partial u_i}{\partial x_j} = -\frac{\partial p}{\partial x_i} + \frac{\partial t_{ij}}{\partial x_j}$$

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## Navier-Stokes equation

$$\rho \frac{\partial u_i}{\partial t} + \rho u_j \frac{\partial u_i}{\partial x_j} = -\frac{\partial p}{\partial x_i} + \frac{\partial t_{ij}}{\partial x_j}$$

## Viscous stress tensor

$$t_{ij} = 2\mu s_{ij}$$

## Strain-rate tensor

$$s_{ij} = \frac{1}{2} \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right)$$

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## Navier-Stokes equation

$$\rho \frac{\partial u_i}{\partial t} + \rho u_j \frac{\partial u_i}{\partial x_j} = -\frac{\partial p}{\partial x_i} + \frac{\partial t_{ij}}{\partial x_j}$$

Unsteady, compressible, threedimensional viscous flow.

## Viscous stress tensor

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## Navier-Stokes equation

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## Mass conservation equation

$$\frac{\partial u_i}{\partial x_i} = 0$$

Incompressible flow.

# Statistical approach

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## Reynolds 1895

### Instantaneous velocity

$$u_i(\mathbf{x}, t) = U_i(\mathbf{x}, t) + u_i'(\mathbf{x}, t)$$

### Reynolds time average

$$U_i(\mathbf{x}, t) = \frac{1}{T} \int_t^{t+T} u_i(\mathbf{x}, t) dt$$

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Reynolds 1895

Instantaneous velocity

$$u_i(\mathbf{x}, t) = U_i(\mathbf{x}, t) + u'_i(\mathbf{x}, t)$$

Reynolds time average

$$U_i(\mathbf{x}, t) = \frac{1}{T} \int_t^{t+T} u_i(\mathbf{x}, t) dt$$

Time averaged equations of motion

$$\frac{\partial U_j}{\partial x_j} = 0$$

$$\rho \frac{\partial U_j}{\partial t} + \rho U_j \frac{\partial U_j}{\partial x_j} = -\frac{\partial P}{\partial x_j} + \frac{\partial}{\partial x_j} \left( 2\mu S_{ij} - \overline{\rho u'_j u'_i} \right)$$

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## Time averaged equations of motion

$$\frac{\partial U_j}{\partial x_j} = 0$$

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## Reynolds-stress tensor

$$\tau_{ij} = -\overline{\rho u'_j u'_i}$$

Fundamental problem of turbulence modeling.

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molecular gradient  
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≈

turbulent stress

$$\mathbf{t}_{ij} = \mu \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right)$$

≈

$$\tau_{ij} = \mu_T \left( \frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right)$$

$\mu$  ... molecular viscosity  
 $\mu_T$  ... eddy viscosity.

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## Reynolds stress tensor

$$\tau_{ij} = \mu_T \left( \frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right)$$

## Eddy viscosity

$$\mu_T = \rho l_{mix}^2 |S_{ij}|$$

Boussinesq eddy-viscosity approximation.

Prandtl's (1925) mixing length hypothesis.

$l_{mix}$  ... mixing length

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- conservation of mass and momentum
- n additional differential transport equations

- Boussinesq eddy viscosity approximation:  $\tau_{ij} = \mu_T \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right)$

Prandtl (1945): eddy viscosity depends upon the kinetic energy of the turbulent fluctuations,  $k$ .

## Turbulence kinetic energy

$$k = \frac{1}{2} \overline{u'_i u'_i}$$

## Eddy viscosity

$$\mu_T = \text{constant} \cdot \rho k^{1/2} l$$

## Trace of Reynolds stress tensor

$$\tau_{ii} = -\overline{\rho u'_i u'_i} = 2\rho k$$

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## Turbulence kinetic energy equation

$$\rho \frac{\partial k}{\partial t} + \rho U_j \frac{\partial k}{\partial x_j} = \tau_{ij} \frac{\partial U_i}{\partial x_j} - \rho \epsilon + \frac{\partial}{\partial x_j} \left[ (\mu + \mu_T / \sigma_k) \frac{\partial k}{\partial x_j} \right]$$

## Turbulence kinetic energy dissipation per unit mass

$$\epsilon = \nu \overline{\frac{\partial u'_i}{\partial x_k} \frac{\partial u'_i}{\partial x_k}}$$

## Reynolds stress tensor

$$\tau_{ij} = 2\mu_T S_{ij} - \frac{2}{3}\rho k \delta_{ij}$$

# One equation models

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## Turbulence kinetic energy dissipation

$$\epsilon \propto k^{3/2} / l$$

Prandtl

## Eddy viscosity

$$\mu_T = \rho k^{1/2} l$$

$l$  - the only unspecified part of the one equation model



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## Turbulence kinetic energy dissipation

$$\epsilon \propto k^{3/2} / l$$

Prandtl

## Eddy viscosity

$$\mu_T = \rho k^{1/2} l$$

$l$  - the only unspecified part of the one equation model

Spalart and Allmaras (1992) - model equations for the eddy viscosity.

## Eddy viscosity

$$\mu_T = \mu_T(\nu, \tilde{\nu})$$

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- Transport equation for turbulence kinetic energy,  $k$ .
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- Transport equation for turbulence kinetic energy,  $k$ .
- Transport equation for dissipation of turbulence kinetic energy,  $\epsilon$ .

Launder and Sharma (1974) - Standard  $k - \epsilon$  model:

Eddy viscosity

$$\mu_T \propto \rho k^2 / \epsilon$$

Turbulence length scale

$$l \propto k^{3/2} / \epsilon$$

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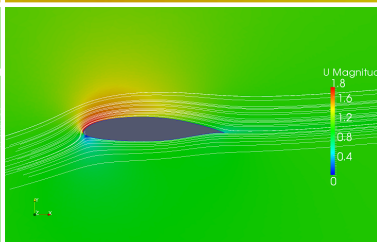
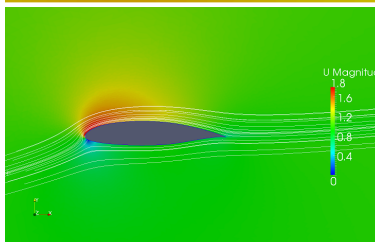
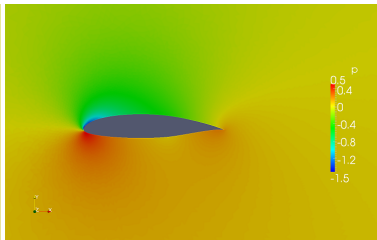
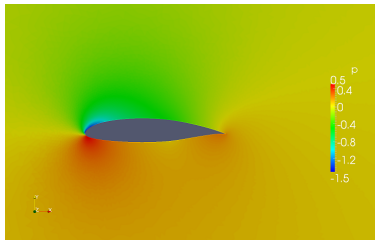
Bibliography

- OpenFOAM - CFD
- simpleFoam - steady, incompressible, viscous, threedimensional flow
- 2D
- $Re = 2.0 \cdot 10^6$

# Airfoil - pressure, velocity magnitude and streamlines

Spalart-Allmaras model

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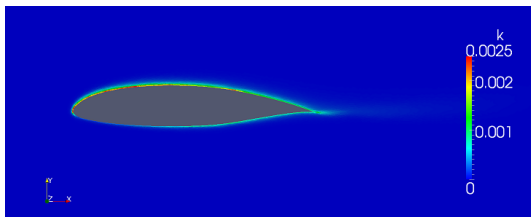
$Re=2.0 \cdot 10^6, \alpha = 10^\circ$

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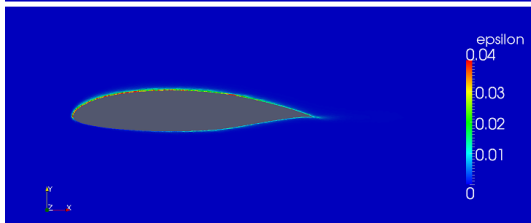
# Airfoil - $k$ and $\epsilon$

$k - \epsilon$  model

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Dissipation of turbulence kinetic energy



$Re=2.0 \cdot 10^6, \alpha = 10^\circ$

# Airfoil - eddy viscosity $\nu_T$

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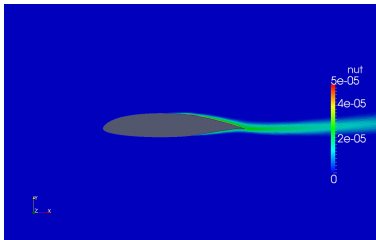
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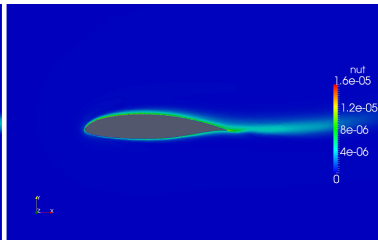
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$Re=2.0 \cdot 10^6, \alpha = 10^\circ$

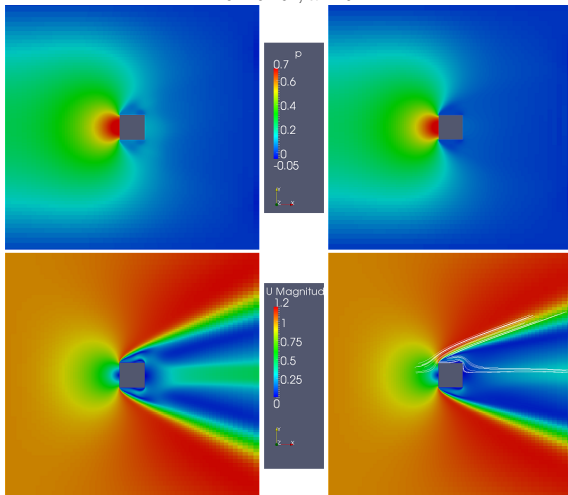


# Square - pressure, velocity magnitude and streamlines

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$Re=2.0 \cdot 10^6, \alpha = 0^\circ$



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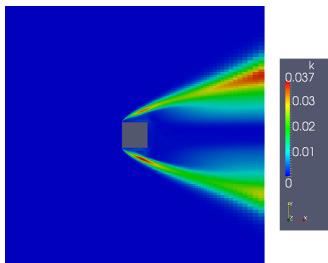
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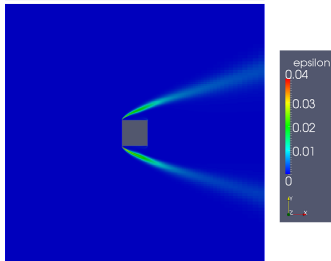
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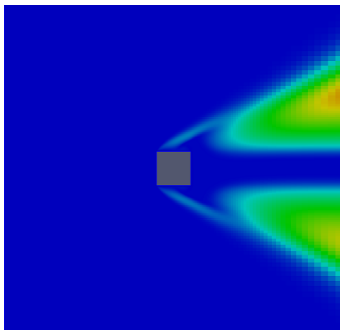
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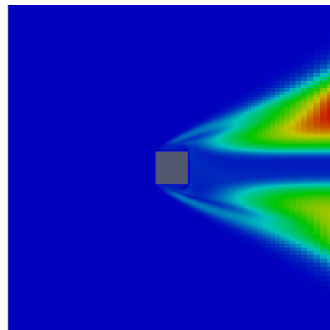
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- Spalart-Allmaras model - fast, quite accurate and stable, excellent for first computation.
- $k - \epsilon$  model - quite accurate, unreliable at large pressure gradients.

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